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Development of High Performance Laminated Electroformed Shape Memory Composite Materials for Lightweight and Deployable Optics

ACT-02-0096
Contract NAS1-03007



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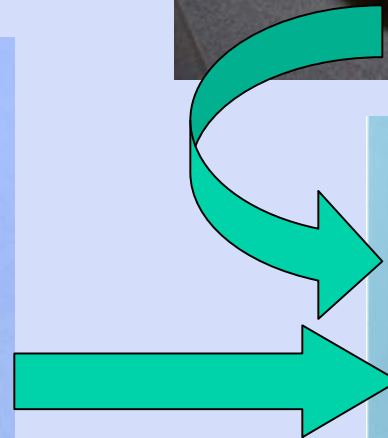
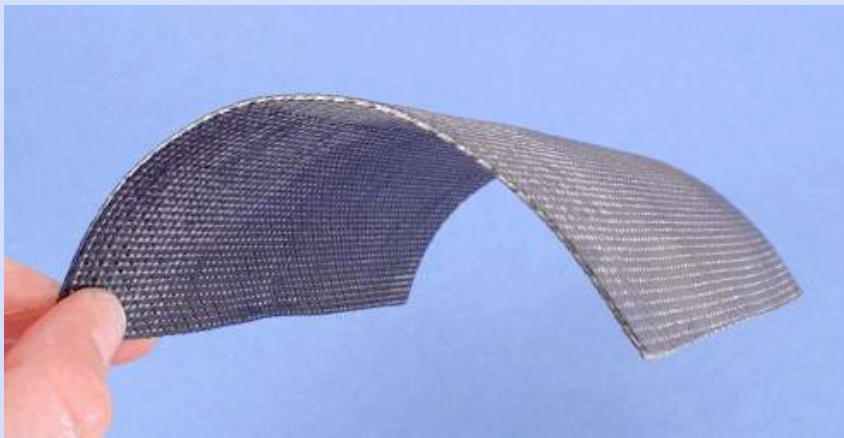
Laminated Deployable Optics



The Best of Two Worlds

Nickel electroformed replica:
Smooth, accurate optical surfaces

Shape memory polymer- carbon fiber
reinforced, for light weight and
controllable deployment



Deployable, lightweight optics



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ESSP Needs for Deployable Optics



- **ESTO Workshop (2003) identified multiple mission needs for deployable imaging systems requiring apertures of 1 to 5 meters and larger**
 - Microwave- soil moisture, temp radiometry, cloud heights
 - IR- temperature measurement, radiometry
 - SAR
 - Visible- Lasercom, LIDAR
- **Deployable optics are an enabling technology for many missions**
 - Lighter weight and stiffer optics
 - Instrument capabilities enhanced at larger apertures
 - Cross enterprise value to OSS, Exploration missions



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Why Consider Shape Memory Composite Optics?



- **Large optics fold in a small package**
- **Replication – Cost and time savings**
- **Larger design parameter space**
- **Adaptable to in-space deployment and active control**
- **Study goals :**
 - Shape control with smooth surface (2 nm RMS), low mass (1-5 kg/m²)
 - Replication accuracy ($<10\lambda$)
 - Stability (temperature, vacuum, creep)
 - Deployability and repeatability (1:10⁴)



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Concepts unique to replicated shape memory/plating laminate



- **Surface Replication: low stress nickel**
 - Replicates optical figure
 - Good surface finish
 - Tough, flexible, Established processes
- **Shape memory resin composite**
 - High stiffness, low mass
 - Suitable for Replication Processes
 - Low outgassing
 - Deployable
- **Manage the interface stress between the composite and nickel for shape control**

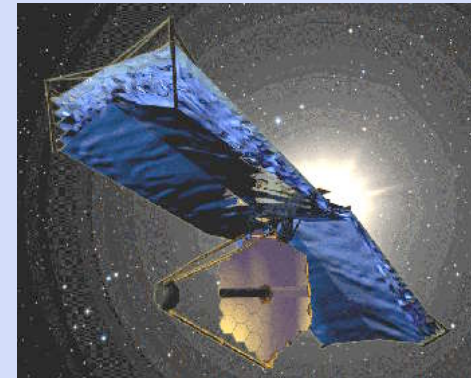


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Team Responsibilities



- **Ball Aerospace**
 - Concept development and evaluation
 - Integrated Modeling
 - Application engineering
- **Cornerstone Research Group**
 - Specialized shape memory polymer tailoring
 - Composite reinforcement schemes
 - Fabrication of composites
- **Northwestern University**
 - Electroplated optics fabrication
 - Materials research expertise





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Combining skills and processes



- **Modeling of the composite structure and nickel**
- **Electroplating processes**
- **Composite resin and structure development**



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Modeling Summary



- **Comparison of model results and experimental results show promise for using the model to help define hardware parameters**
- **Current simplified model will be translated into more detailed FEM and thermal models**
- **These models can be easily added to an integrated end-to-end model being developed under other applications**



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Model Goals and Assumptions



- **Development of material performance models for design**
 - Modeling based on measured material properties
 - Provide information for future sample preparation
 - Characterization and definition of properties/processes/geometry to characterize or control
- **Continue to update design models using measured performance data and define parameters necessary for optimizing and scaling-up the process to larger optics**
- **Current model is simple, Excel based**
- **Can be used to “engineer” the needed materials and structures**
 - Preferential shaping, balancing thickness and properties
- **Incorporates thermal, material and structural properties**
 - Input parameters defined by test results
- **Includes flat and single curved mirrors**
- **Each material layer is homogeneous**



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Preliminary Results of Model



- **Model run for Styrene SMP with 3D Weave Carbon**
- **Deflections (“roll-up”) of convex samples within 50% of experimental data**
 - Due to temperature change (110 C in oven)
 - Due to added forces
 - Formula used for flat samples to be modified for better accuracy on curved samples
- **Change in length of flat samples due to temperature change within 20% of experimental data**



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Detailed Modeling



- **Update for new material constructions**
 - Symmetric laminations for stability and uniformity
- **Comparison with strain measurements made on samples**
 - New constructions fabricated and entering test
- **Scalability**
 - Utilize scaling methods developed for other large space structures
- **Translate current model into more detailed Finite Element and thermal models that are linked**
 - Include double convex design
 - Include other materials in the lamination
 - These models can eventually be integrated into a formal Integrated End-to-End Modeling (IM) environment
- **IM can include theoretical models for items such as the effect of material microyield on the mirror**

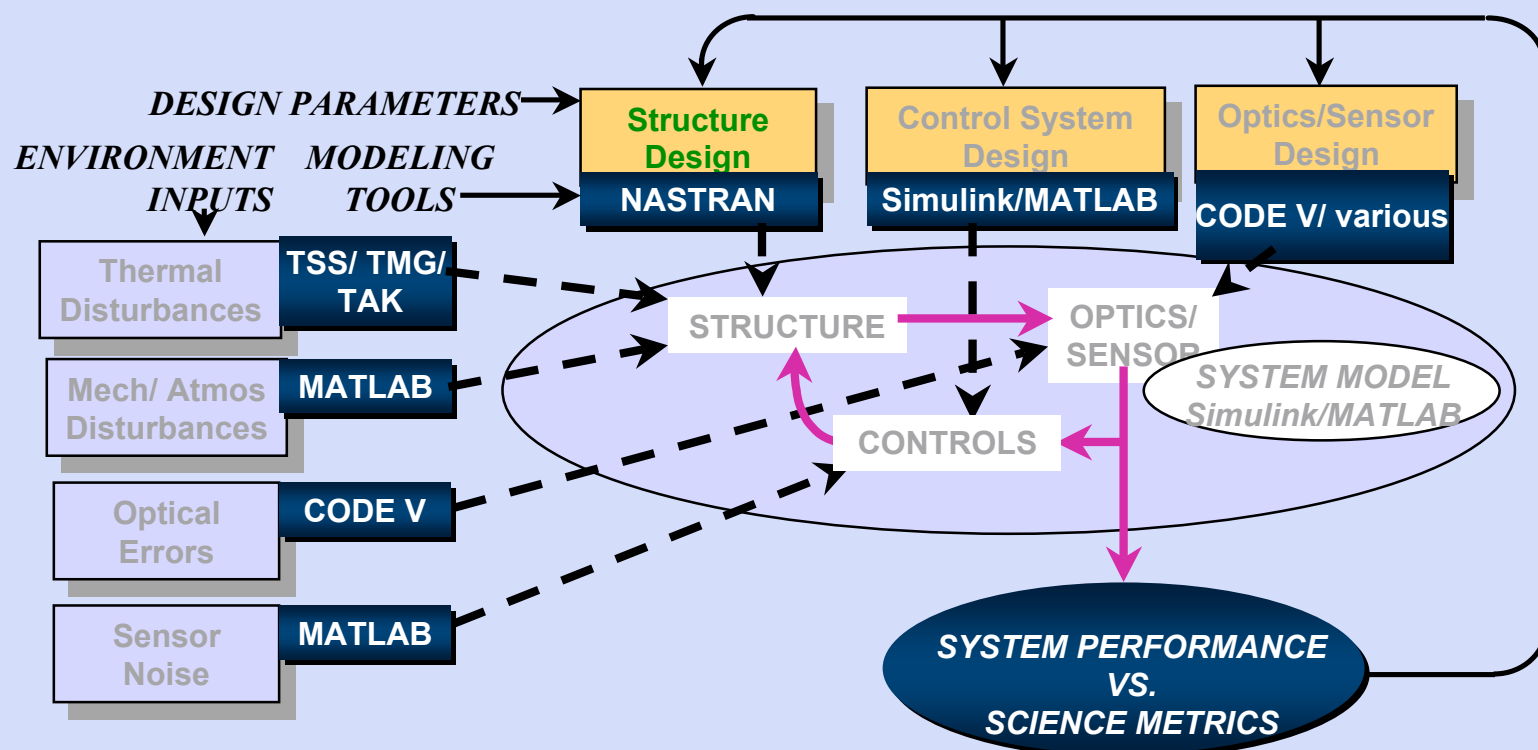


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Ball Integrated End-to-End Modeling Environment



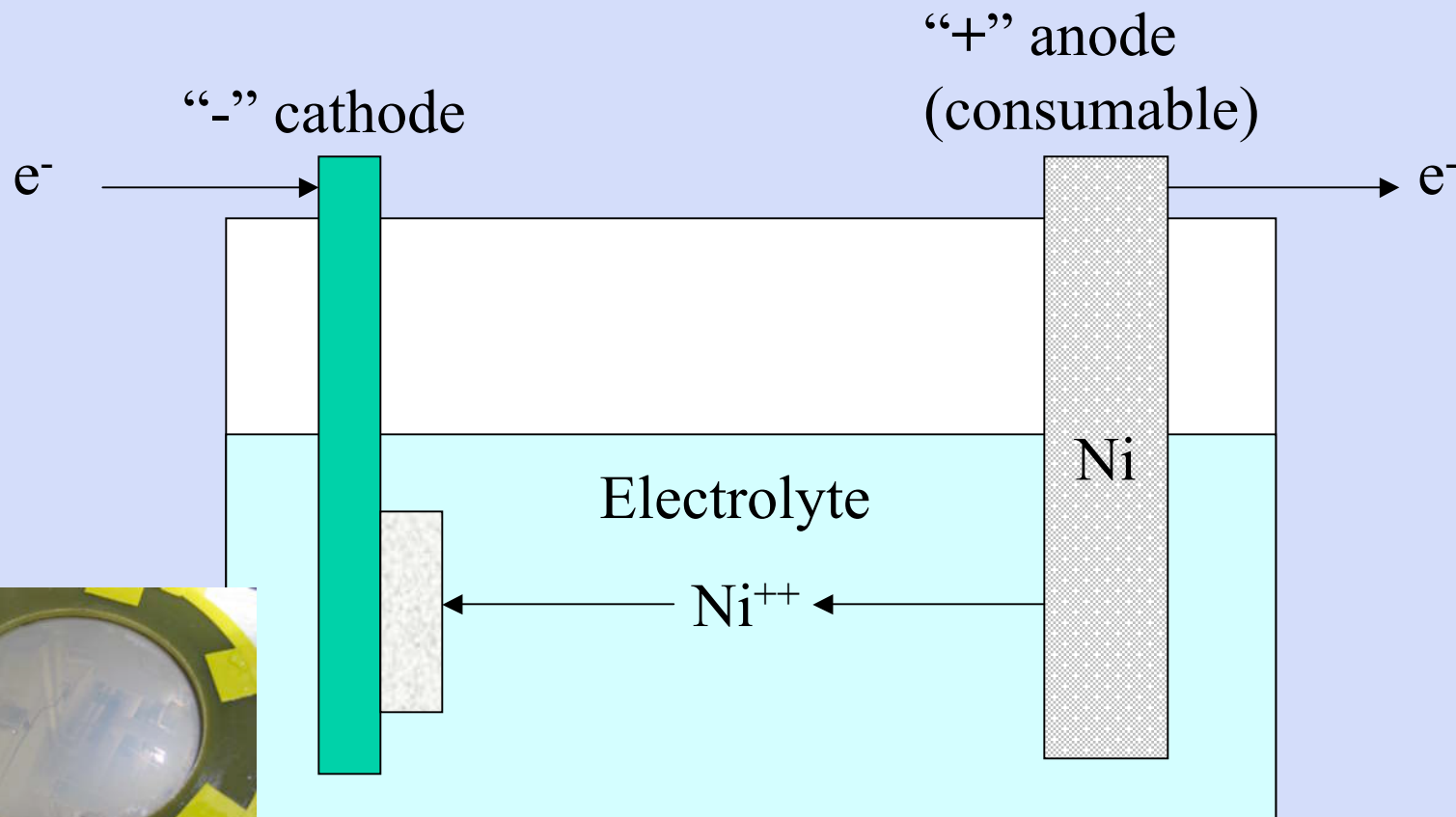
- Unique Ball capability under development for 7 years.
- State of the art integration capability for coupled /optical /structural /controls /sensors / signal processing/ and disturbance models - no artificial boundaries between disciplines.
- Simulink / MATLAB based environment - cradle to grave system engineering tool
- Note that experts retain traditional role.





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Low Stress Electroplating





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Two Approaches for Mirror Fabrication



Electrofabrication (replication)

Conductive and/or
Release Layer

Replica

Master/Mandrel

Reflective Surface

Electroplating

Reflective Surface

Conductive Layer
(for Non-conductive
Substrate)

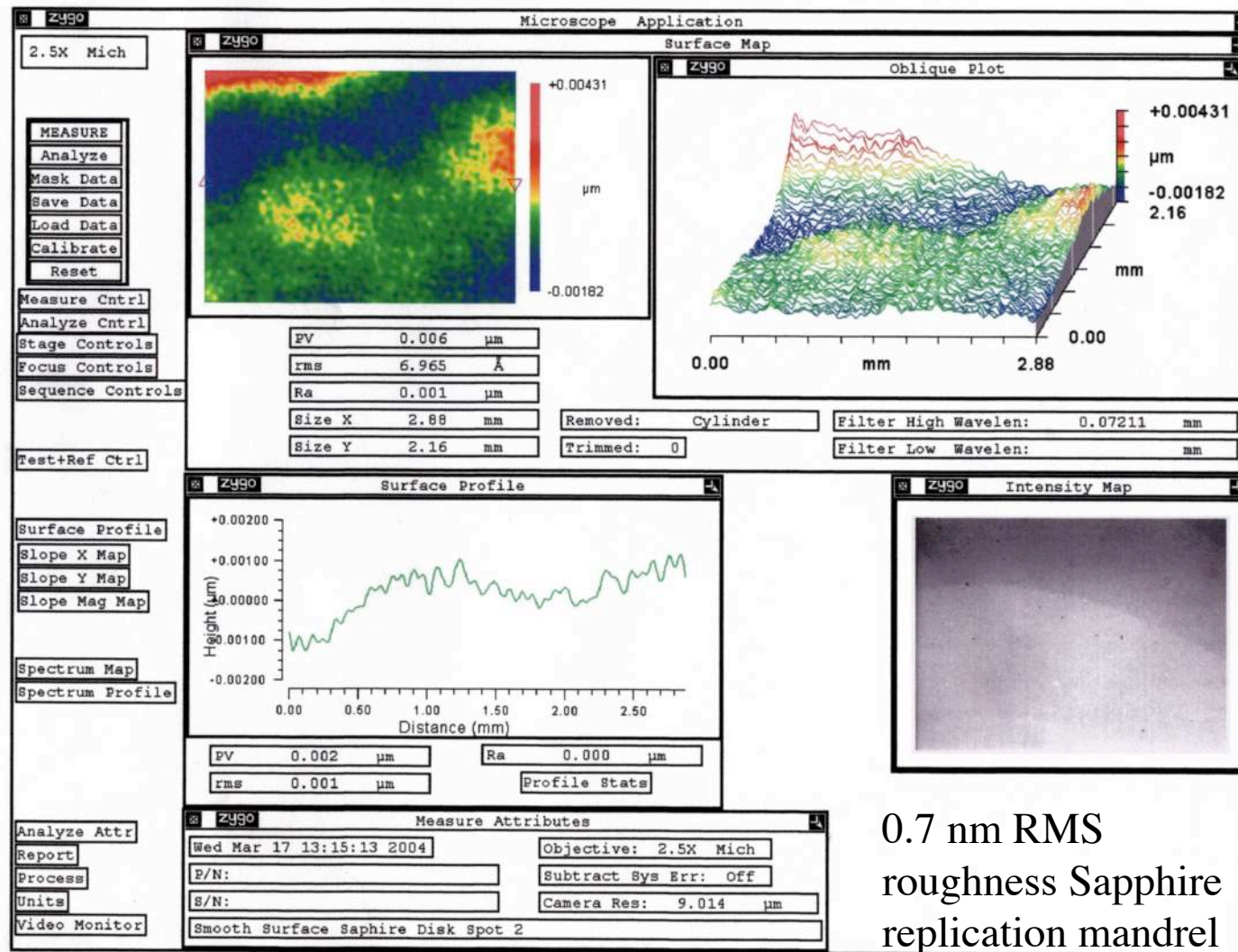
Bright Ni

Substrate



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Superpolish Surface Replicated in Nickel

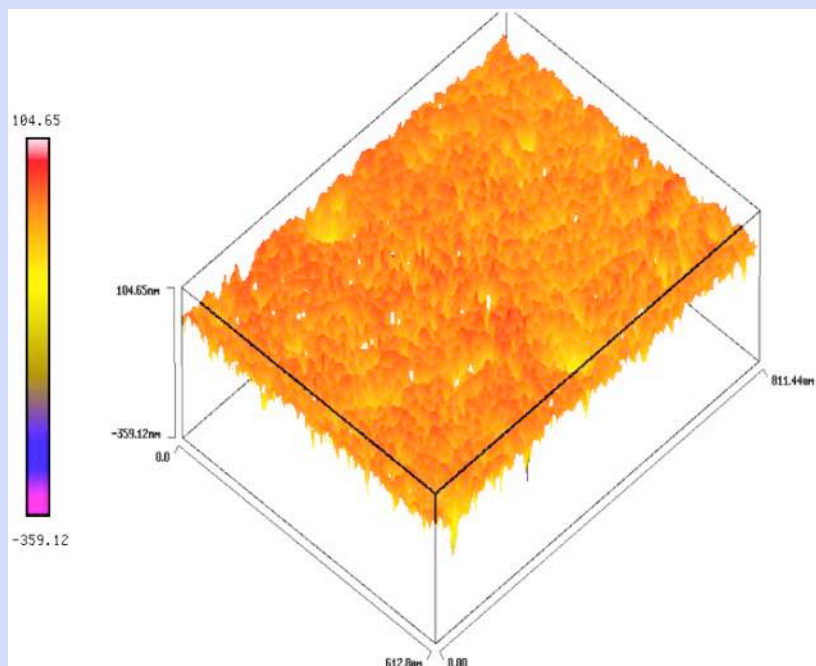


0.7 nm RMS
roughness Sapphire
replication mandrel



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Surface roughness data meets 2nm RMS



- Zygo interferometer scan data from a developmental cyanate ester resin cast sample.
- Sample shows resin itself meets 7nm roughness, and is unlikely to distort nickel

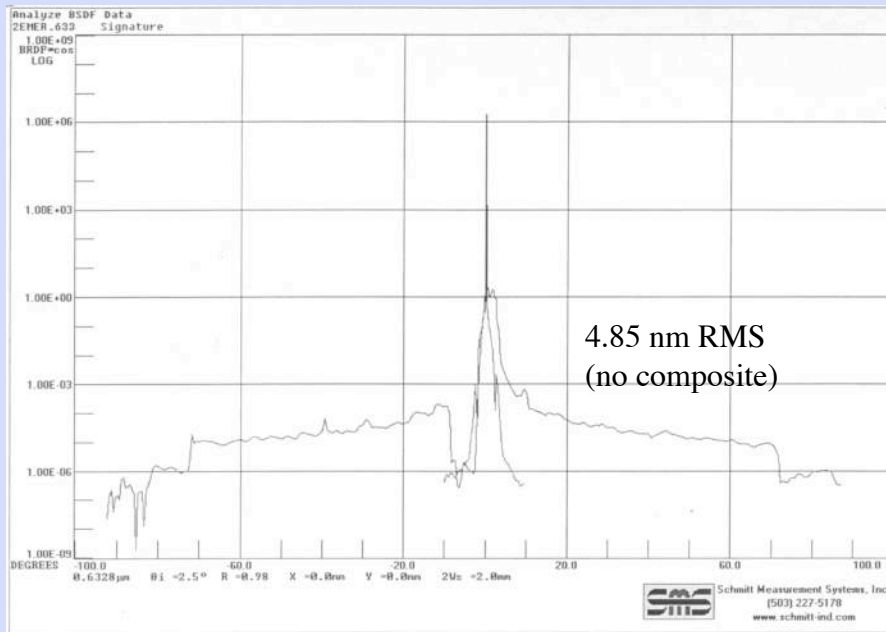
- Zygo interferometer scan data shows nickel can meet at least 2 nm roughness
- 0.7 nm polish sapphire flats coated to demonstrate maximum replication smoothness





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Surface scatter replication dominated by print through



Print through of fabric weave,
about 100 nm RMS (O = 1 mm)

- 4.85 nm RMS surface roughness replicated in 20 micron thick nickel from polished flat (no composite backing)
- After laminating composite to the back of 20 micron nickel, print-through results in high apparent roughness, profilometer indicates 3 microinch (100 nm) roughness about ~ 33 nm RMS measured with scatterometer
- New composite replication process has demonstrated <5 nm RMS roughness, to be tested with high smoothness (0.3nm) nickel replication mandrel



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Shape Memory Polymers Application



- **Actuation**
 - Store and release mechanical energy
- **Reconfiguration**
 - Temporary modulus reduction to enable shape change
- **Replication**
 - Replicate surface from master in manufacturing processes

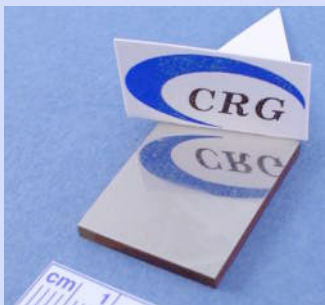


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SMP Development - Cyanate Ester



- **CE polymers already used in space**
- **Transformed conventional CE to SMP**
 - Fully cured, cross-linked for stability
 - Required new polymer design
- **CE shape memory polymer results:**
 - Deformation-recovery cycle demonstrated
 - Activation temperature of 160°C
- **Optimization**
 - Enhanced strain recovery
 - Increased toughness

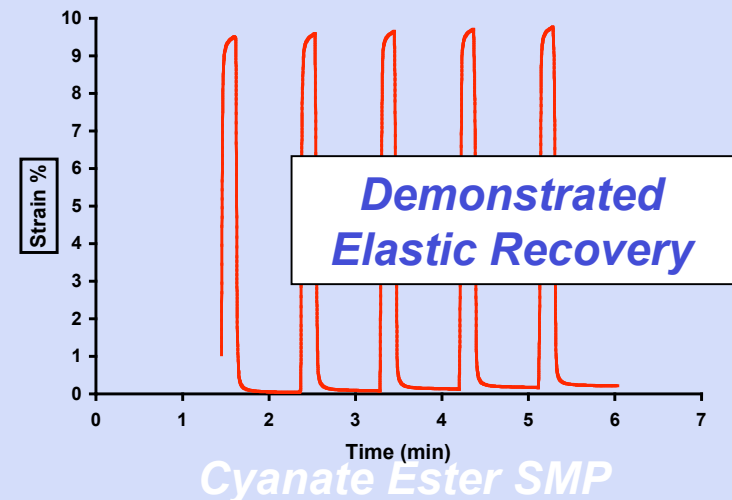
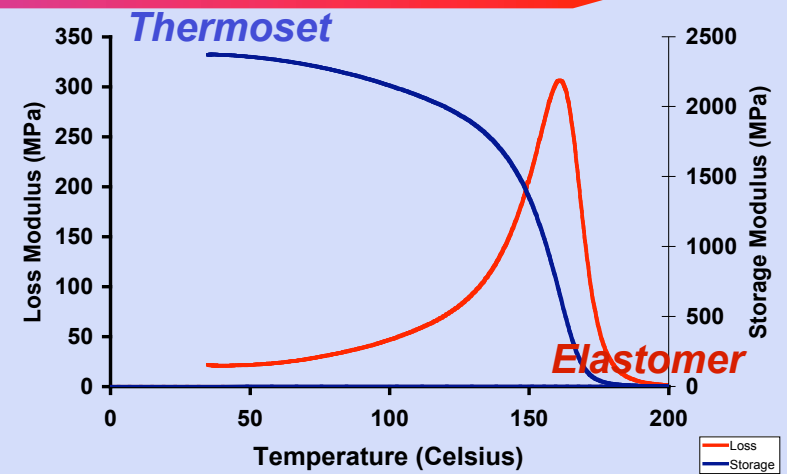


Coated CE SMP



Carbon Nanofiber-CE SMP Composite Membrane (0.5.kg/m²)

Cyanate Ester SMP Modulus





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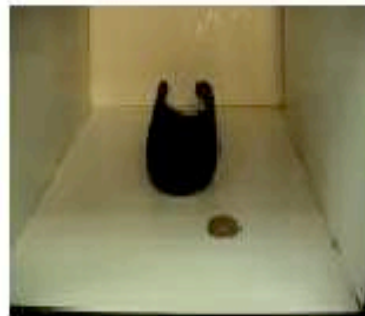
Deployment test of spherical surface -no damage to structure



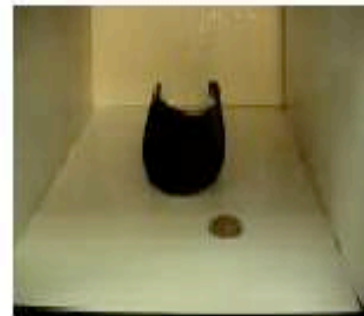
- Self-Deployment tested in oven at 110C
- Lower temp activation polymer being developed



000s



025s



038s



053s



067s



087s



134s



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Initial Deployment Test Results Mixed



Before rolling

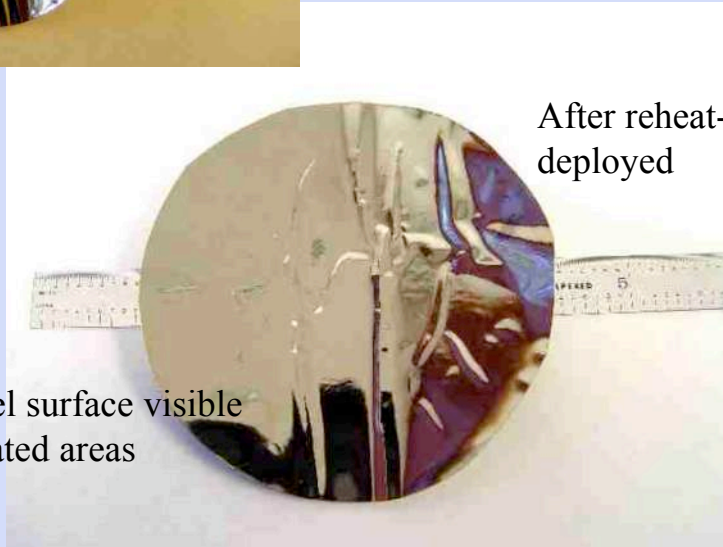


Rolled to 15mm radius (stow)



Delamination occurred during
heating, not during rolling

After reheat-
deployed



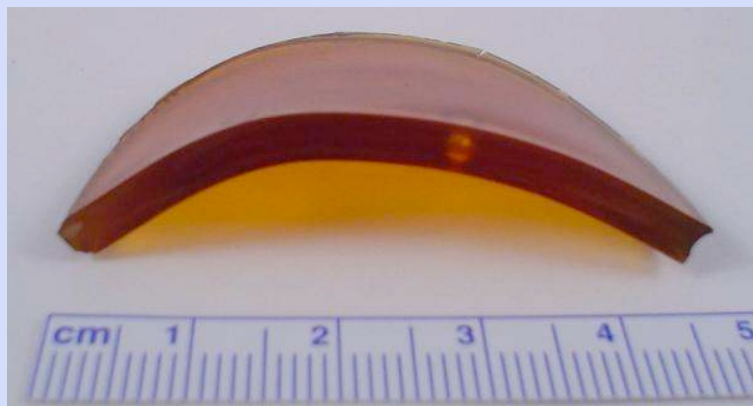
Damage to nickel surface visible
only in delaminated areas

- **First Deployment Test**
 - Optical figure after release from mandrel was poor- astigmatic
 - Temperature cycling – shape was unchanged by heating, some additional delamination
 - Deployability concept validated— Areas without delamination did not show nickel surface damage when stowed at a 15 mm radius (100x tighter than a flight item) and then redeployed to near-original shape
 - Nickel-composite adhesion will be researched further - several options available

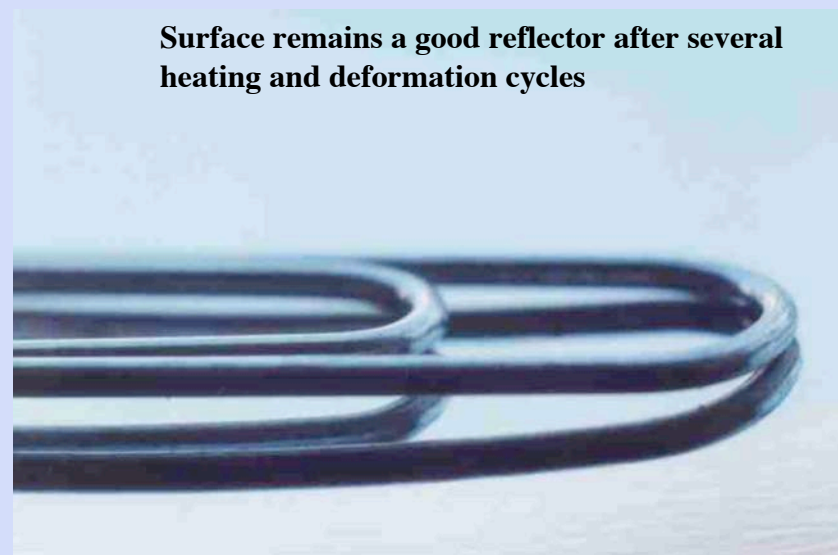


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Deployed Mirror surface undamaged by deployment



- Buckling was observed but only where delaminated
- Remains smooth after several heat annealing cycles
- Surface is smooth at resin cure temp (CTE mismatch)





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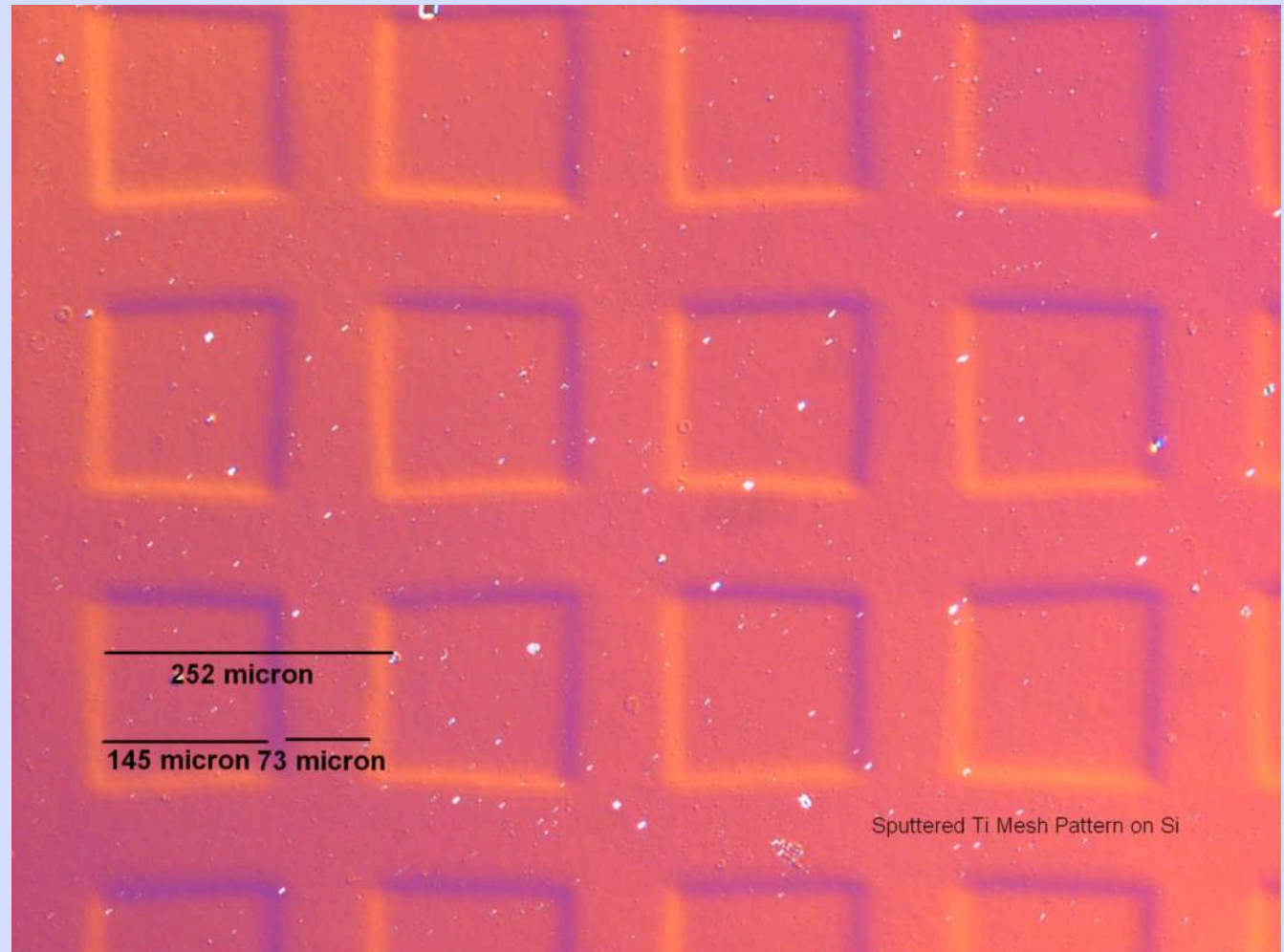
Ion etched rear surface of nickel greatly improves adhesion



No
Delamination

Thermal cycled
0-200C

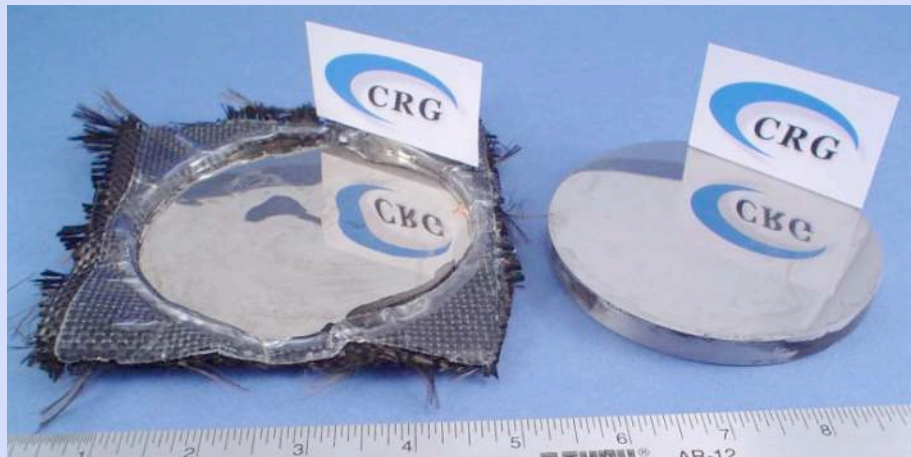
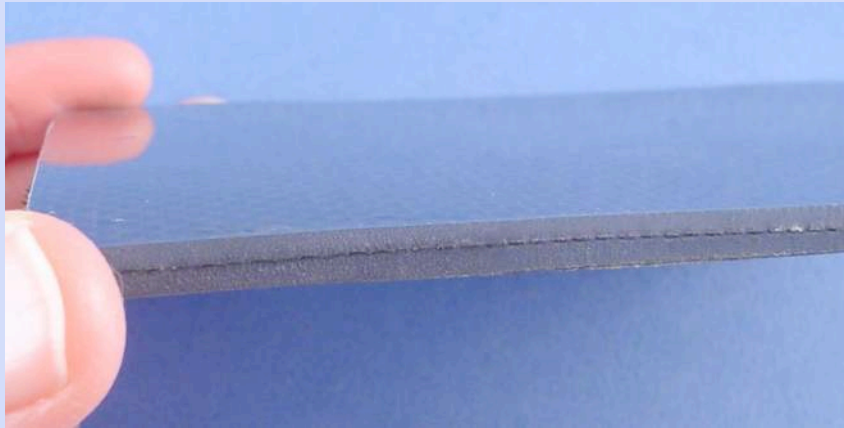
Deformed to
20 mm radius





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Print through remains a challenge, but is significantly reduced



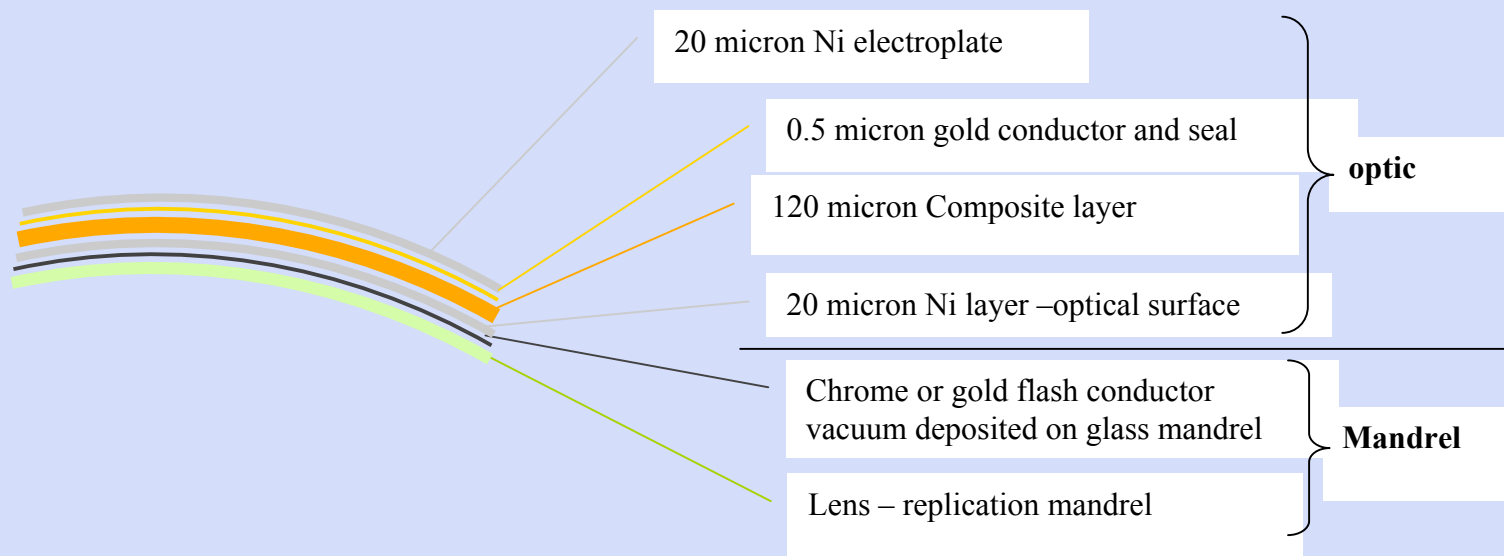
- Fiber-resin CTE mismatch produces print through
 - Sandwich approach is improvement, without fibers near surface
 - Neat (resin rich) layer surrounds fiber reinforcement
 - Nanofibers and alternate filler reinforcement being investigated
-
- Neat resin layer reduces fiber print-through effect, but nickel-thermal mismatch caused waviness



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MultiLayer Lamination

reduced moisture loss 40%



- **Nickel on both optical surface and rear surface equalizes thermal stresses and seals composite from moisture absorption/loss and outgassing creep**
- **Thermal vacuum tests show 40% lower moisture loss. Stress relief and dimensional stability TBD**
- **Optical testing -figure and surface roughness poor on this sample due to abandoned structure concept. Test to be repeated on a better test item utilizing new adhesion process and structure for print through reduction.**
- **Deployment test –Determine if it can be rolled and unrolled without permanent damage with delamination and creasing.**



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Shape Memory Mirror Feasibility Demonstrated



- **Figure**: Low stress nickel process produces <10 waves PTP mirror. Composite replication needs development
- **Roughness**: Low scatter nickel achieved 2nm RMS
- **Outgassing**: Shape Memory CE resin meets requirements (0.16% TML)
- **Spherical Surfaces**: Reinforced composite SMP applied to flat and spherical nickel plating surfaces
- **Adhesion**: Demonstrated ruggedness of nickel-composite lamination
- **Stow and deployment** demonstrated without damage to optical surface, deployment repeatability needs development



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Critical parameters for continued investigation



- **Modeling** –Validate prediction of symmetric layers and measured composite properties. Stress model to include matrix of position elements to translate to surface figure and optical performance. Validate scaling laws.
- **Flat mirror production** fabricate test items for structural data and model verification, and surface finish/print through
- **Spherical mirror production** –compare performance of improved adhesion, and reduced print through, improved surface figure, figure adjustment
- **Deployability**: cycling without damage or change, repeatability
- **Stability and creep**: improvement on conventional composites
- **Polymer development**:
 - low CTE resin/composite
 - low cure temp minimizes
 - Reinforcement of resin for CTE match, low print through structure
- **Optimize nickel thickness to match composite properties**



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Near Term Goals



- **Scale up- to 30-40 cm**
 - Complete analytical modeling of components and lamination and fold into the integrated model for optical predictions
 - Validate analytical model and extend to spherical surfaces
 - Demonstrate optical surface replication and stability
 - Demonstrate repeatability of deployment (overcome residual strain)
- **Develop structural concepts for accurate deployment**
- **Validate key processes and performance,
and the connection between them**
 - Low temp cure and deployment,
 - Reverse side plating for stability
 - Evaluate low CTE resin/composite and room temp cure



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Conclusions



- **Laminated nickel – composite optics can provide deployable optical surfaces without delamination or deterioration of metal surface and substrate**
- **Shape memory control provides more design latitude for deployable optics**
- **Typical deployment accuracy is consistent with needs for microwave reflectors of 1-2 meter diameter**
- **Additional work needs to be done**
 - Resin chemistry (low CTE, low temp cure,)
 - Structures (optimizing deployment, modeling in stow condition)
 - Durability in the space environment